





# Frame statics solved using CALFEM toolbox for MATLAB

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### Approximation

$$\mathbf{u}^{e}(x) = \mathbf{N}^{e}(x)\mathbf{q}^{e}$$
$$\mathbf{N}^{e} = \begin{bmatrix} L_{i}^{e} & 0 & 0 & L_{j}^{e} & 0 & 0 \\ 0 & H_{i}^{e} & \hat{H}_{i}^{e} & 0 & H_{j}^{e} & \hat{H}_{j}^{e} \end{bmatrix}, \mathbf{q}^{e} = \begin{bmatrix} q_{1} \\ q_{2} \\ q_{3} \\ q_{4} \\ q_{5} \\ q_{6} \end{bmatrix}$$











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Displacements vector

 $\mathbf{u} = \{u(x), v(x)\}$ 

#### Matrix of constitutive relationships

$$\mathbf{D} = \left[ \begin{array}{cc} EA & 0\\ 0 & EI \end{array} \right]$$

#### Strain vector

 $\mathbf{e} = \{\varepsilon_x, \kappa\}$ 

Stress vector

 $\mathbf{s} = \{N(x), M(x)\}$ 

Distributed loading

$$\mathbf{p} = \{p_x, p_y\}$$

Differential operator matrix

$$\mathbf{L} = \begin{bmatrix} \frac{\mathrm{d}}{\mathrm{d}x} & 0\\ 0 & -\frac{\mathrm{d}^2}{\mathrm{d}x^2} \end{bmatrix}$$

#### Kinematic and constitutive relations

$$\mathbf{e} = \mathbf{L}\mathbf{u} = \mathbf{L}\mathbf{N}\mathbf{q} = \mathbf{B}\mathbf{q}, \qquad \mathbf{s} = \mathbf{D}\mathbf{e} = \mathbf{D}\mathbf{B}\mathbf{q}$$







#### Element stiffness matrix

$$\mathbf{k}^{e} = \int_{0}^{l^{e}} \mathbf{B}^{e^{T}} \mathbf{D}^{e} \mathbf{B}^{e} dx^{e}$$
$$\mathbf{k}^{e} = \frac{EI}{l^{3}} \begin{bmatrix} \frac{Al^{2}}{I} & 0 & 0 & -\frac{Al^{2}}{I} & 0 & 0\\ 0 & 12 & 6l & 0 & -12 & 6l\\ 0 & 6l & 4l^{2} & 0 & -6l & 2l^{2}\\ -\frac{Al^{2}}{I} & 0 & 0 & \frac{Al^{2}}{I} & 0 & 0\\ 0 & -12 & -6l & 0 & 12 & -6l\\ 0 & 6l & 2l^{2} & 0 & -6l & 4l^{2} \end{bmatrix}^{e}$$

Transformation matrix  $c = \cos(\alpha^e)$  i  $s = \sin(\alpha^e)$ 

$$\mathbf{T}^{e} = \begin{bmatrix} c & s & 0 & 0 & 0 & 0 \\ -s & c & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & c & s & 0 \\ 0 & 0 & 0 & -s & c & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix}$$







#### Substitute nodal forces

$$\mathbf{z}^{e} = \int_{0}^{l^{e}} \mathbf{N}^{e^{\mathrm{T}}} \mathbf{p}^{e} \mathrm{d}x^{e}$$
  
for  $p_{x} = \mathrm{const} = \widehat{p}_{x}$  i  $p_{y} = \mathrm{const} = \widehat{p}_{y}$   
 $\mathbf{z}^{e} = \left\{ \frac{\widehat{p}_{x}l}{2}, \ \frac{\widehat{p}_{y}l}{2}, \ \frac{\widehat{p}_{y}l^{2}}{12}, \ \frac{\widehat{p}_{x}l}{2}, \ \frac{\widehat{p}_{y}l}{2}, \ -\frac{\widehat{p}_{y}l^{2}}{12} \right\}$ 









# Assignment









# Discretization









#### function frame()









function frame()

% definition of dof matrix % for elements Edof=[1 1 2 3 4 5 6 ; 2 4 5 6 7 8 9; 3 7 8 9 10 11 12];









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% compute coordinate vectors % for elements [Ex,Ey]=coordxtr(Edof,Coord,Dof,2);









% material/section properties E=2e7; I=1e-4; A=0.03; ep=[E,A,I];









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% zero global matrices K=zeros(12); F=zeros(12,1);

% account for % concentrated loads F(4)=100;F(9)=50;









### Equivalent nodal forces









% plot the frame eldraw2(Ex,Ey, [1,2,2]);







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% compute stiffness matrices for elements Ke1=beam2e(Ex(1,:),Ey(1,:),ep); Ke2=beam2e(Ex(2,:),Ey(2,:),ep); [Ke3,Ze3]=beam2e(Ex(3,:),Ey(3,:),ep,eq);







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% compute stiffness matrices for elements Ke1=beam2e(Ex(1,:),Ey(1,:),ep); Ke2=beam2e(Ex(2,:),Ey(2,:),ep); [Ke3,Ze3]=beam2e(Ex(3,:),Ey(3,:),ep,eq);

% assemble global stiffness matrix and load vector K=assem(Edof(1,:),K,Ke1); K=assem(Edof(2,:),K,Ke2); [K,F]=assem(Edof(3,:),K,Ke3,F,Ze3);















% compute displacement and reaction force vector [Q,R]=solveq(K,F,bc)







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% extract nodal displacements for all elements Qe=extract(Edof,Q);







% compute displacement and reaction force vector [Q,R ]=solveq(K,F,bc)

% extract nodal displacements for all elements Qe=extract(Edof,Q);

% return to elements to compute nodal forces f1=beam2s(Ex(1,:),Ey(1,:),ep,Qe(1,:)) f2=beam2s(Ex(2,:),Ey(2,:),ep,Qe(2,:)) f3=beam2s(Ex(3,:),Ey(3,:),ep,Qe(3,:),eq)







% compute displacement and reaction force vector [Q,R ]=solveq(K,F,bc)

% extract nodal displacements for all elements  $Qe{=}extract(Edof,Q);$ 

% return to elements to compute nodal forces f1=beam2s(Ex(1,:),Ey(1,:),ep,Qe(1,:)) f2=beam2s(Ex(2,:),Ey(2,:),ep,Qe(2,:)) f3=beam2s(Ex(3,:),Ey(3,:),ep,Qe(3,:),eq)

% draw deformed frame eldisp2(Ex,Ey,Qe,[1,4,1]);







## Results

Q=	R=	f1=		
0	-69.3890	43.7836	-53.8986	-73.2148
0	-2.6878	43.7836	-53.8986	61.5316
0	73.2148			
0.0355	0.0000	f2=		
-0.0264	-0.0000			
-0.0073	-0.0000	-16.2164	26.1014	61.5316
0.0003	0.0000	-16.2164	26.1014	-3.7219
-0.0001	0.0000			
0.0288	0.0000	f3=		
0	-30.6110			
0	32.6878	-20.5168	-22.8761	-53.7219
0	15.6586	-44.5168	-4.8761	15.6586







# Force sign convention in FE









% return to elements to compute nodal forces f1=beam2s(Ex(1,:),Ey(1,:),ep,Qe(1,:),[0,0],7) f2=beam2s(Ex(2,:),Ey(2,:),ep,Qe(2,:),[0,0],7) f3=beam2s(Ex(3,:),Ey(3,:),ep,Qe(3,:),eq,21)





```
% return to elements to compute nodal forces
f1=beam2s(Ex(1,:),Ey(1,:),ep,Qe(1,:),[0,0],7)
f2=beam2s(Ex(2,:),Ey(2,:),ep,Qe(2,:),[0,0],7)
f3=beam2s(Ex(3,:),Ey(3,:),ep,Qe(3,:),eq,21)
```

```
% Deformed frame
figure(1)
eldraw2(Ex,Ey, [1,2,2]);
eldisp2(Ex,Ey,Qe,[1,4,1]);
axis([-1 7 -1 5]);
title('Displacements')
plotpar=[2 1];
```







```
% Normal forces
figure(2)
scal=scalfact2(Ex(3,:),Ey(3,:),f3(:,1),0.35);
eldia2(Ex(1,:),Ey(1,:),f1(:,1),plotpar,scal);
eldia2(Ex(2,:),Ey(2,:),f2(:,1),plotpar,scal);
eldia2(Ex(3,:),Ey(3,:),f3(:,1),plotpar,scal);
axis([-1 7 -1 5])
title('Normal forces')
```





```
% Normal forces
figure(2)
scal=scalfact2(Ex(3,:),Ey(3,:),f3(:,1),0.35);
eldia2(E_{x}(1,:),E_{y}(1,:),f1(:,1),plotpar,scal);
eldia2(Ex(2,:),Ey(2,:),f2(:,1),plotpar,scal);
eldia2(Ex(3,:),Ey(3,:),f3(:,1),plotpar,scal);
axis([-1 7 -1 5])
title('Normal forces')
% Shear forces
figure(3)
scal=scalfact2(Ex(1,:),Ey(1,:),f1(:,2),0.35);
eldia2(Ex(1,:),Ey(1,:),f1(:,2),plotpar,scal);
eldia2(Ex(2,:),Ey(2,:),f2(:,2),plotpar,scal);
eldia2(Ex(3,:),Ey(3,:),f3(:,2),plotpar,scal);
axis([-1 7 -1 5]);
title('Shear forces')
```







```
% Moments
figure(4)
scal=scalfact2(Ex(1,:),Ey(1,:),f1(:,3),0.35);
eldia2(Ex(1,:),Ey(1,:),f1(:,3),plotpar,scal);
eldia2(Ex(2,:),Ey(2,:),f2(:,3),plotpar,scal);
eldia2(Ex(3,:),Ey(3,:),f3(:,3),plotpar,scal);
axis([-1 7 -1 5]);
title('Bending moments');
```





# Diagrams



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